

## FORMULARIO ASTRODINAMICA

**Orbite ellittiche:  $0 < e < 1$**

$$r = \frac{h^2}{\mu} \frac{1}{1 + e \cos \theta} \quad a = \frac{h^2}{\mu} \frac{1}{1 - e^2}$$

$$M_e = E - e \sin E$$

$$T = 2\pi \sqrt{\frac{a^3}{\mu}}$$

$$\tan \frac{E}{2} = \sqrt{\frac{1-e}{1+e}} \tan \frac{\theta}{2}$$

$$M_e = \frac{\mu^r}{h^3} (1 - e^r)^{(3/2)} t = \frac{2\pi}{T} t = nt$$

**Orbite iperboliche:  $e > 1$**

$$r = \frac{h^r}{\mu} \frac{1}{1 + e \cos \theta} \quad a = \frac{h^2}{\mu} \frac{1}{e^2 - 1} \quad e = 1 + \frac{r_p v_\infty^r}{\mu_{planet}}$$

$$M_h = \frac{\mu^2}{h^3} (e^2 - 1)^{(3/2)} t$$

$$\tanh \frac{F}{2} = \sqrt{\frac{e-1}{e+1}} \tan \frac{\theta}{2}$$

$$M_h = e \sinh F - F$$

$$v_\infty = \frac{\mu}{h} e \sin \theta_\infty = \frac{\mu}{h} \sqrt{e^r - 1} \quad v_{bo} = v_p = \frac{h}{r_p} = \sqrt{v_\infty^2 + \frac{2\mu}{r_p}}$$

$$\beta = \cos^{-1}(1/e)$$

$$\theta_\infty = \cos^{-1}(-1/e)$$

**Cambio piano**

$$\Delta v = \sqrt{v_1^2 + v_2^2 - 2v_1v_2[\cos \Delta \gamma - \cos \gamma_2 \cos \gamma_1(1 - \cos \delta)]}$$

$$\cos i = \cos \phi \sin A$$

**Flyby**

$$\phi_2 = \phi_1 + \delta \quad \text{dark side}$$

$$\phi_2 = \phi_1 - \delta \quad \text{sunlit side}$$

**Clohesy-Wiltshire**

$$\Phi_{rr}(t) = \begin{bmatrix} 4 - 3\cos(nt) & 0 & 0 \\ 6(\sin(nt) - nt) & 1 & 0 \\ 0 & 0 & \cos(nt) \end{bmatrix} \quad \Phi_{rv}(t) = \begin{bmatrix} \sin(nt)/n & 2(1 - \cos(nt))/n & 0 \\ 2(\cos(nt) - 1)/n & 1(4\sin(nt) - 3nt)/n & 0 \\ 0 & 0 & 1 \cos(nt)/n \end{bmatrix}$$

$$\Phi_{vr}(t) = \begin{bmatrix} 3n\sin(nt) & 0 & 0 \\ 6n(\cos(nt) - 1) & 0 & 0 \\ 0 & 0 & -n\sin(nt) \end{bmatrix} \quad \Phi_{vv}(t) = \begin{bmatrix} \cos(nt) & 2\sin(nt) & 0 \\ -2\sin(nt) & 4\cos(nt) - 3 & 0 \\ 0 & 0 & \cos(nt) \end{bmatrix}$$

**Trasformazione di coordinate:**

$${}^{Orb}_{Geo}R = \begin{bmatrix} \cos \Omega \cos \omega - \sin \Omega \sin \omega \cos i & \sin \Omega \cos \omega + \cos \Omega \cos i \sin \omega & \sin i \sin \omega \\ -\cos \Omega \sin \omega - \sin \Omega \cos i \cos \omega & -\sin \Omega \sin \omega + \cos \Omega \cos i \cos \omega & \sin i \cos \omega \\ \sin \Omega \sin \omega & \sin i \cos \omega & \cos i \end{bmatrix} \quad {}^{Orb}_{Geo}R = {}^{Geo}_{Orb}R^T$$

**Regresione linea dei nodi:**

$$\dot{\Omega} = - \left[ \frac{3}{2} \frac{\sqrt{\mu} J_2 R^2}{(1 - e^2)^2 a^{\frac{7}{2}}} \cos i \right] \quad \Delta \phi_1 = \frac{-2\pi\tau}{\tau_e} \quad \Delta \phi_2 = \dot{\Omega} \tau \quad n|\Delta \phi| = m2\pi$$

$$\text{sunsincronismo } \dot{\Omega} = 1.991 * 10^{-7} \frac{rad}{sec} \quad \dot{\omega} = - \left[ \frac{3}{2} \frac{\sqrt{\mu} J_2 R^2}{(1 - e^2)^2 a^{\frac{7}{2}}} \right] \left( \frac{5}{2} \sin^2 i - 2 \right)$$